1 **Uniform needle-emitter arrays for ionic liquid electrospray thrusters**

2 **with precise thrust control**

3

4 Fumiya Tachibana[1], Toshiyuki Tsuchiya[2], and Yoshinori Takao[3*]

5

6 *[1]Department of Mechanical Engineering, Materials Science, and Ocean Engineering,*

7 *Yokohama National University, 79-5 Tokiwadai, Hodogaya-ku, Yokohama, Kanagawa 240-*

8 *8501, Japan*

9 *[2]Department of Micro Engineering, Kyoto University, C3 Bldg., Kyoto University-Katsura,*

10 *Nishikyo-ku, Kyoto 615-8540, Japan*

11 *[3]Division of Systems Research, Yokohama National University, 79-5 Tokiwadai, Hodogaya-ku,*

12 *Yokohama, Kanagawa 240-8501, Japan*

13

14 *E-mail: takao@ynu.ac.jp

15

16 **Abstract**

17 The development of ionic liquid electrospray thrusters with highly precise needle-emitter arrays

18 is reported. Micro-electro-mechanical systems process technology is applied in the fabrication

19 process of needle-emitter arrays to achieve a uniform shape of emitter tips. The resulting

20 emitter-array chips were then tested to gather the current–voltage characteristics of emitter

21   arrays with different numbers of emitters. The maximum ion current extracted from the emitters

22   increased almost in proportion with the number of emitters. When the current–voltage curves

23   of 81-, 169-, and 361-emitter chips were compared with a constant gap distance between the

24   emitter and extractor electrodes, the onset voltage of ion emission was nearly constant because

25   the emitter tips on all the chips were uniform in shape. Moreover, the current–voltage curves

26   had similar slopes for the different number of emitters after the onset voltage, which

27   demonstrates the uniform ion current output of all the emitter arrays.

28

1. Introduction

The number of launched nano/microsatellites has been increasing since 2013; more than 300 nano/microsatellites were launched in 2017.[1] Because nano/microsatellites can be developed quickly and at a low cost,[2] small companies and universities as well as large companies and government agencies can make and operate nano/microsatellites.[3–10] These satellites (< 100 kg) are now deployed for diverse applications, and advanced missions, such as formation flights and constellations of nano/microsatellites, are now being planned and carried out.[11–13] Such nano/microsatellite based missions require propulsion systems that can be mounted on nano/microsatellites and achieve precise thrust control to maintain the relative positions of a satellite formation.[14–16]

Ionic liquid electrospray thrusters hold great potential as miniature propulsion systems with precise thrust control for nano/microsatellites because they neither require a gas propellant nor a neutralizer, which drastically reduces the size of the entire propulsion system compared to conventional alternatives. Figure 1 shows a schematic of a general electrospray thruster. The electrospray thruster comprises needle-shaped emitters and an extractor electrode. By applying a high voltage of about a few kilovolts between the emitters and the extractor electrode, the ionic liquid, which is the propellant of the thruster, flows to the emitters' tips. Because the curvature radius of the emitter tips is only a few micrometers, a strong electric field is generated at the emitter tip. This strong electric field deforms the ionic liquid into a conical structure known as a Taylor cone.[17] When the force of the electric field is stronger than the surface

3

49 tension pressure of the ionic liquid, ions evaporate directly from the surface. The extracted ions

50 are then accelerated by the potential difference between the emitters and the extractor electrode

51 to produce thrust.[18,19]

52   Ionic liquids are molten salts that consist of only positive and negative ions (cations

53 and anions), and their vapor pressure is nearly zero because of the strong Coulomb forces

54 between the ions.[20] Coulomb interactions allow ionic liquids to remain in the liquid phase even

55 in vacuum, making it easier to store the propellant in a compact package as compared with

56 electric propulsion systems that use gas propellants, e.g., ion thrusters. In addition, high-

57 pressure gas feed systems, which require mechanical valves for gas flow control can be

58 eliminated in electrospray thrusters, thus eliminating mechanical vibrations from the system

59 and further enhancing the precision of thrust control. Moreover, electrospray thrusters can

60 extract cations and anions by applying a positive or negative voltage to the emitters,

61 respectively. This feature eliminates the need for neutralizers as cations and anions can be

62 extracted alternately with bipolar pulse voltage, or both the ions can be extracted simultaneously

63 using a pair of electrospray thrusters.[21,22] Although neutralizers cannot yield thrust owing to

64 the very small mass of electrons, anions can generate thrust because anions and cations have

65 similar molecular weights. Furthermore, because each emitter of an electrospray thruster

66 produces only a small amount of thrust (of the order of 10 nN), the thrust can be controlled

67 precisely by adjusting the number of emitters and the voltage applied to the emitters.

68   In our previous study, we fabricated externally-wetted needle-shaped emitters by using

4

69 micro-electro-mechanical systems (MEMS) processing and we conducted ion current

70 measurements to develop ionic liquid electrospray thrusters.[23)] The resulting maximum

71 extracted ion current with four emitters was about 8 µA at an applied voltage of 3700 V, and

72 the thrust was estimated to be 0.77 µN. However, the extracted ion current hardly increased

73 with increase in the number of emitters because the emitter tips were nonuniform.

74 This lack of uniformity meant that the number of emitters in an array could not be used

75 to control the ion current. The present study addresses this issue with the fabrication of a

76 uniform emitter array and conduction of ion emission experiments that use the emitter arrays

77 in electrospray thrusters. Here, we report a MEMS process that yields arrays with uniform

78 emitter tips and the current–voltage characteristics of emitter chips with different numbers of

79 emitters.

80 The rest of the paper is organized as follows. In Sect. 2, we describe the fabrication

81 process and the ion emission experiment of emitter chips with one and nine emitters. Because

82 the ion current did not increase with the number of emitters in tests with these small chips, the

83 fabrication process was revised to achieve better uniformity in the emitter tips. This revised

84 process and test results with 81-, 169-, and 361-emitter chips are then discussed in Sect. 3.

85 Section 4 draws conclusions to this study.

86

87 **2. Fabrication and current measurements of single and nine-emitter arrays**

88 First, we fabricated emitter chips with one and nine emitters to compare the ion current

89   output. These preliminary tests with small numbers of emitters were designed to clarify whether

90   the ion current would depend on the number of emitters with the MEMS fabrication process.

91   2.1. Fabrication process

92   We fabricated the emitter chips using the steps that are illustrated in Fig. 2. First, we

93   coated a silicon wafer with a thick positive-type photoresist (PMER P-LA900PM). This

94   positive-type photoresist was transcribed from a photomask pattern by using photolithography.

95   Second, we performed isotropic etching using SF_6 to form the cone shape of the emitter tips,

96   based on the patterned photoresist layer. Then, the ionic liquid reservoir was formed by the

97   Bosch process,[24,25] after which the photoresist was removed. Figure 3 shows the fabricated

98   chips with a single emitter and an array of nine emitters. Both emitter chips are 10 mm × 10

99   mm in dimension, and the central region (8 mm in diameter) forms the ionic liquid reservoir,

100   which encompasses the emitters. For the single emitter, the diameters of the emitter base and

101   tip are 100 µm and 6 µm, respectively. The nine emitters in the array are aligned in a straight

102   line with the pitch of 500 µm, and all the emitter bases are 100 µm in diameter. However, as

103   shown in Figs. 3(e) and 3(f), the emitter-tip diameters came out different; the tip diameter at

104   the center of the array is 12 µm, while the emitter at the left end is only 1 µm in diameter.

105   These differences in tip diameter were caused by nonuniform isotropic etching.[23] In

106   the isotropic etching process shown in Fig. 2(b), the etching rate increased in proportion to the

107   distance from the center of the emitter array because of local variations in the pattern density

108   and total exposed area.[26] In addition, the etching rate varied with position within a single silicon

109 wafer. Here, we fabricated four emitter chips on a 4-inch wafer: three emitter chips with a single

110 emitter and one emitter chip with an array of nine emitters, and the etching rate increased in

111 proportion to the distance from the center of the wafer owing to non-uniform plasma density in

112 the process chamber. The non-uniform etching rate, depending on the position within the array

113 and the wafer, was responsible for the nonuniform diameters of the emitter tips.

114 2.2. Experimental setup

115 We tested the fabricated emitter chips in ion emission experiments. Figure 4(a) shows

116 a schematic of the experimental setup that we used for ion current measurements. The distance

117 between the emitter tip and extractor was set at $d = 0.3$ mm, and the collector electrode (18 mm

118 $\times$ 30 mm) was placed at a distance of $L = 14$ mm from the extractor electrode. Most of the

119 systems under test (emitter chip, extractor, and collector) were in a vacuum chamber evacuated

120 by a rotary pump and a turbomolecular pump at a base pressure $<1.0 \times 10^{-3}$ Pa. Figures 4(b)

121 and 4(c) show the extractor electrodes for the single emitter and the array of nine emitters,

122 respectively. The extractor electrodes are 0.08-mm thick metal plates, and contain an aperture

123 that is 0.5 mm in diameter for the single emitter and has 0.5 mm $\times$ 5 mm slit for the array of

124 nine emitters. These extractor electrodes were aligned with the emitter chip using the same

125 alignment system employed in our previous paper,[23] with which alignment must be conducted

126 manually and examined visually. Thus, by merging a row of apertures into a slit, we can easily

127 align the emitter and extractor and reduce the number of ions intercepted by the extractor,

128 although the electric field at the emitter tips will be slightly reduced. We used an ionic liquid

129   1-ethyl-3-methylimidazolium dicyanamide (EMI-DCA) as a propellant for testing, and placed

130   a small drop of ionic liquid (0.01 μL) on the reservoir to be supplied to the emitter. The emitter

131   chip was biased from 0 to 3000 V using a source meter (Keithley 2657 A) connected through a

132   resistor of 1 MΩ. The extractor and the collector electrodes were connected to different source

133   meters (Keithley 237) through 100 kΩ and 1 MΩ resistors, respectively. The output voltage to

134   the extractor and collector electrodes was set at 0 V (equivalent to the ground potential) to

135   facilitate the detection of ions colliding against the electrodes.

136   2.3. Results and discussion

137   Figure 5 shows the results of ion emission experiments with a single emitter and an

138   array of nine emitters. From the single emitter, ions were extracted at an applied voltage ranging

139   from 2200 to 2300 V, and the maximum extracted ion current was 0.61 μA. From the nine-

140   emitter array, ions were extracted at an applied voltage ranging from 1600 to 2600 V, and only

141   1.4 μA of ion current was extracted at an applied voltage of 2600 V. Note that electrical short

142   circuits between the emitter and extractor occurred in both cases over the applied voltage shown

143   in the figure. These results indicate that the ion current did not increase in proportion to the

144   number of emitters, which is explained by the large difference in the tip diameters of the nine

145   emitters. The onset voltage of ion emission, V_{start}, can be estimated from the following equation:

$$V_{\text{start}} = \sqrt{\frac{\gamma R_{\text{c}}}{\varepsilon_0}} \ln\left(\frac{4d}{R_{\text{c}}}\right) \tag{1}$$

146   where γ is the surface tension of the ionic liquid, R_{c} is the curvature radius of the ionic liquid,

147   which has a paraboloidal shape at the emitter tip, ε_0 is the dielectric constant of vacuum, and d

148　is the distance between the emitters and the extractor electrode.[27] This equation indicates that

149　the onset voltage of ion emission depends significantly on the radius of curvature at the emitter

150　tip. Therefore, emitter tips with different radii of curvature will have different onset voltages,

151　and no ions will be extracted from emitters with relatively large diameter tips. In addition,

152　because the shapes of the tips on the single emitter and nine-emitter array were irregular, the

153　onset voltage and slopes of current–voltage curves were not consistent. We report this initial

154　fabrication process to clarify the design choices we made in a second attempt with a revised

155　approach, which is reported below.

156

157　**3. Revised fabrication and current measurements**

158　　　We revised the fabrication process to obtain emitter arrays with uniform emitter tips

159　because the nonuniformity in the emitter tips makes precise control of the extracted ion current

160　difficult, whether the applied voltage or the number of emitters is used to control the ion current.

161　3.1. Revised fabrication process

162　　　Figure 6 shows the revised emitter chip fabrication process. We added two steps to the

163　fabrication process illustrated in Fig. 2. First, the diameter of the resist used as the mask for

164　isotropic etching was adjusted. In the isotropic etching process [Fig. 6(c)], the etching rate

165　increased in proportion to the distance from the center of the emitter array or the wafer. Such a

166　non-uniform etching rate resulted in the non-uniform tip diameters. Therefore, we obtained an

167　isotropic etching rate for each emitter depending on its position, and the mask diameter was

9

168   adjusted for each tip so that the resulting tips would be uniform in diameter. Second, we applied

169   a second isotropic etching step [Fig. 6(f)] after removing the resist [Fig. 6(e)]. We found that

170   the etching rates were nearly uniform in the second isotropic etching step. Because the

171   difference in etching rate was caused by local variations in the exposed area,[26] uniform

172   isotropic etching was achieved in the second etching step, for which the exposed area was

173   uniform. Moreover, we reduced the time of the first isotropic etching step that led to the

174   nonuniform etching rate by dividing the isotropic etching process into the two steps shown in

175   Figs. 6(c) and 6(f). The shorter time of the first isotropic etching step minimizes differences in

176   the mask diameter, which allowed us to reduce the differences in the diameters of emitter bases.

177   The second isotropic etching was intended to sharpen the emitter tips without the risk of

178   overetching in the first etching step; overetching often causes the loss of the mask before the

179   anisotropic etching step [Fig. 6(d)], thereby destroying the emitters.

180   Figure 7(a) shows the photograph of an emitter chip fabricated using the

181   abovementioned revised process. The emitter chip is 18 mm × 18 mm in dimension and

182   encompasses an array of 361 (19 × 19) emitters fabricated within a 13×13 mm^2 area in the

183   center of the chip with a pitch of 500 µm. Figure 7(b) shows the emitter at the center of the

184   array, and Figs. 7(c)–7(e) show four emitters at different positions in the array. Figures 8 and 9

185   show the similarly fabricated emitter chips with 169 (13 × 13) and 81 (9 × 9) emitters,

186   respectively. For the 169- and 81-emitter chips, the pitch of the emitters was 500 µm; the

187   emitters at the outermost row of the emitter array had different base diameters from the rest of

188   the emitters, depending on the mask diameters. As mentioned earlier, the mask diameters were

189   adjusted in proportion to the etching rate of the first isotropic etching, and the emitter-base

190   diameters were determined by the mask diameter used in the Bosch process. Although the base

191   diameter of some emitters was different, as expected, we successfully obtained uniform and

192   sharp emitter tips, regardless of their positions from the center of the array.

193   3.2. Experimental setup

194   The experimental setup for the ion current measurement was almost the same as that

195   described in Sect. 2.2, but the distance between the emitter tip and the extractor was set to $d =$

196   0.4 mm to avoid electrical short circuits, and a 70 mm $\times$ 70 mm collector electrode was placed

197   at a distance of $L = 30$ mm from the extractor electrode. In addition, we applied a 1-Hz bipolar

198   pulse voltage to the emitters. For ionic liquid ion sources, an electrochemical double layer is

199   formed between the emitters and the ionic liquid.[28] Because this double layer is very thin, if

200   charge accumulation causes even a small potential difference across the gap between layers,

201   the resulting strong electric field may corrode the emitters and ionic liquid. Charge

202   accumulation across this double layer happens in the order of ~10 s; therefore, a voltage applied

203   at 1-Hz alternating frequency prevents corrosion.[29] Figure 10(a) shows the extractor electrode,

204   which was used for all the arrays (81-, 169-, and 361-emitter chips). The extractor electrode

205   was fabricated from a 525-μm thick silicon wafer using the Bosch process. Figure 10(b) shows

206   a schematic cross-sectional view of the extractor electrode. The 11 mm $\times$ 11 mm square groove

207   etched on the wafer surface reduces the thickness of the apertures from 525 to 125 μm. The

208   diameter of apertures is 300 μm, and the pitch is 500 μm.

209   We used EMI-DCA as the propellant, and a drop of ionic liquid (0.1 μL) was placed

210   on the reservoir in the same way as in Sect. 2. Figure 11 shows an assembled test device. The

211   emitter chip and the extractor electrode were aligned using microbeads, and they were placed

212   into a polyether ether ketone fixture. As shown in Fig. 11(b), electrical contacts for the emitter

213   and the extractor were connected through the upper fixture holes.

214   3.3. Results and discussion

215   Figure 12 shows the results of ion emission experiments with 81-, 169-, and 361-

216   emitter chips. For the 81-emitter chip, the maximum ion current was 22.8 μA on the positive

217   side and the minimum was −19.5 μA on the negative side. For the 169-emitter chip, the

218   maximum ion current was 42.0 μA on the positive side and −36.5 μA on the negative side. For

219   the 361-emitter chip, the maximum ion current was 96.7 μA on the positive side and −87.6 μA

220   on the negative side. These results indicate that the maximum ion current extracted from the

221   emitters increased almost in proportion to the number of emitters, and the maximum ion current

222   was much larger than what we observed previously.[23]

223   For all three emitter chips, the slopes of the current–voltage curves decreased at an

224   applied voltage of ~±2400 V. This transition indicates that the current–voltage characteristics

225   changed around these points. The change in voltage dependence around ±2400 V could have

226   arisen from the insufficient flow of ionic liquid to the emitter tips. In a previous study, when

227   high voltage was applied, the emission current from each emitter was limited by the supply of

228 ionic liquid to emission sites owing to the high hydraulic impedance of the flow paths along the

229 emitter surface.[30] The limited flow rate to the emission sites at high voltages likely explains

230 the transition in the slope of the current–voltage curves.

231   The current intercepted by the extractor electrode was measured at 1% of the emitter

232 current immediately after the ion emission onset voltage and increased linearly until it reached

233 15% at the applied voltage of ±2500 V. In the applied voltage range ±(2500–3000) V, the

234 interception fraction was stable at 14%–16%. In addition, no electrical short circuits were

235 observed between the emitter chip and the extractor electrode in any of the emitter chips. These

236 results indicate that the ion emission was stable even when a high voltage was applied. However,

237 a nonnegligible amount of ions was intercepted by the extractor. This rate could be reduced by

238 optimizing the electrode configuration with adjustments to the emitter shapes, extractor

239 aperture diameters, and gap distances. We leave this optimization for future work.

240   The onset voltage of ion emission was almost equal for all three emitter chips on both

241 the positive and negative sides: 2040 V on the positive side and −2040 V on the negative side

242 for the 81-emitter chip, 2010 V on the positive side and −2010 V on the negative side for the

243 169-emitter chip, and 2000 V on the positive side and −2010 V on the negative side for the 361-

244 emitter chip; the difference in onset voltage is $\leq 2\%$. From Eq. (1), one can see that our

245 observation of nearly equal onset voltages at the fixed gap distance $d = 0.4$ mm implies that

246 differences in the curvature radii of the emitter tips among the three emitter chips were

247 negligible.

248   Figure 13 plots the ion current per emitter for 81-, 169-, and 361-emitter chips as a

249   function of applied voltage. The current per emitter at ±2400 V was 170 nA on the positive side

250   and −160 nA on the negative side with 361-emitter chip, 168 nA on the positive side and −155

251   nA on the negative side with 169-emitter chip, and 180 nA on the positive side and −165 nA

252   on the negative side with 81-emitter chip. The differences in these current-per-emitter

253   measurements for three emitter chips at the same applied voltage were small on both the

254   positive and negative sides. Moreover, the slopes of the current–voltage curves on both the

255   positive and negative sides for the different numbers of emitters are similar. This result

256   demonstrates that all the emitters were emitting ions uniformly, which is explained by the

257   uniform-shaped tips forming uniform electric fields around the emitters. These results

258   emphasize that the shape of the emitter tips is a critical factor in ionic liquid electrospray

259   thrusters, and that the ion current can be controlled with the number of emitters or the voltage

260   to the emitters.

261

262   **4. Conclusions**

263   Toward the development of ionic liquid electrospray thrusters that can be mounted on

264   nano/microsatellites to achieve precise thrust control, we have fabricated and tested needle-

265   emitter arrays from silicon wafers. In preliminary tests, we fabricated single emitter and nine-

266   emitter chips to confirm that ions could be extracted from the emitters. However, the extracted

267   ion current did not increase with the number of emitters because the emitter tips were irregular

268 in shape, which led to inconsistencies in the current–voltage curves of the emitter chips. To

269 improve the uniformity of the emitter tips, we divided the isotropic etching process into two

270 steps in a revised fabrication process, which was successful.

271   Ion emission tests showed that the maximum ion current extracted from the revised

272 emitters increased in proportion to the number of emitters on both the positive and negative

273 sides, and emitter chips with various numbers of emitters had almost the same emission onset

274 voltage. This result indicates that all the emitter tips had a uniform radius of curvature. In

275 addition, the ion currents per emitter were same at the same applied voltage and the slopes of

276 the current–voltage curves were similar for all chips with different number of emitters. This

277 result emphasizes that the shape of the emitter tips is critical to the success of such a thruster

278 and further demonstrates that the emitter arrays we fabricated operate uniformly.

279   In the future work, we plan to measure the ion beam characteristics of emitter arrays

280 fabricated with our process, such as the composition and energy distribution of the ion beam,

281 which we expect to comprise monomers, dimers, trimers, or droplets depending on the

282 conditions.[31,32] Because the mass-to-charge ratio of ions or droplets has a significant effect on

283 the resulting thrust and specific impulse,[33] beam composition will need to be investigated by

284 time-of-flight measurements and the energy distribution will need to be measured using a

285 retarding potential analyzer to estimate thruster performance.[34]

286

287 **Acknowledgments**

15

References

1) SpaceWorks, 2020 Nano/Microsatellite Marcket Forecast, 2020.

2) H. Heidt, J. Puig-Suari, A. S. Moore, S. Nakasuka, and R. J. Twiggs, 14th the AIAA/USU Conf. Small Satellites, 2000, SSC00-V-5.

3) D. Lev, R. M. Myers, K. M. Lemmer, J. Kolbeck, H. Koizumi, and K. Polzin, Acta Astronaut. **159**, 213(2019).

4) D. C. Alhorn, J. P. Casas, E. F. Agasid, C. L. Adams, G. Laue, C. Kitts, and S. O'brien, 25th the AIAA/USU Conf. Small Satellites, 2011, SSC11-VI-1.

5) T. Wekerle, J. B. P. Filho, L. E. V. L. Costa, and L. G. Trabasso, J. Aerosp. Technol. Manag. **7**, 269 (2017).

6) T. Toizumi, T. Enomoto, Y. Yatsu, T. Nakamori, N. Kawai, K. Ishizaka, A. Muta, H. Morishita, K. Akiyama, N. Kisa, S. Inagawa, M. Kawakubo, J. Nishida, S. Mizunuma, S. Matsunaga, and J. Kataoka, Physica E **43**, 685 (2011).

7) S. Rucinski, K. Carroll, R. Kuschnig, J. Matthews, and P. Stibrany, Adv. Space Res. **31**, 371 (2003).

8) H. Koizumi, K. Komurasaki, J. Aoyama, and K. Yamaguchi, Trans. Jpn. Soc. Aeronaut. Space. Sci., Aerosp. Technol. Jpn. **12**, Tb_19 (2014).

9) H. Koizumi, H. Kawahara, K. Yaginuma, J. Asakawa, Y. Nakagawa, Y. Nakamura, S. Kojima, T. Matsuguma, R. Funase, J. Nakatsuka, and K. Komurasaki, Trans. Jpn. Soc. Aeronaut. Space. Sci., Aerosp. Technol. Jpn. **14**, Pb_13 (2016).

10) J. Asakawa, H. Koizumi, K. Nishii, N. Takeda, M. Murohara, R. Funase, and K. Komurasaki, Trans. Jpn. Soc. Aeronaut. Space. Sci., Aerosp. Technol. Jpn. **16**, 427 (2018).

11) S. Bandyopadhyay, R. Foust, G. P. Subramanian, S. Chung, and F. Y. Hadaegh, J. Spacecr. Rockets **53**, 567 (2016).

12) G. Bonin, N. Roth, S. Armitage, J. Newman, B. Risi, and R. E. Zee, 29th the AIAA/USU Conf. Small Satellites, 2015, SSC15-I-4.

13) R. Pirayesh, A. Naseri, R. Adcock, S. Stochaj, N. Shah, and J. Krizmanic, 2018 SpaceOps Conference, AIAA Paper-2633.

14) P. Song, L. Sun, S. Kuang, K. Zhang, W. Zou, X. Suo, Y. Wang, D. Xiao, L. Tu, 36th Int. Electric Propulsion Conf., 2019, IEPC-Paper-A-284.

15) M. M. Micci and A. D. Ketsdever, *Micropropulsion for Small Spacecraft* (American Institute of Aeronautics

17

321   and Astronautics, Reston, VA, 2000).

322   16) C. Scharlemann, N. Buldrini, R. Killinger, M. Jentsch, A. Polli, L. Ceruti, L. Serafini, D. DiCara, and D.

323   Nicolini, Acta Astronautica **69**, 822 (2011).

324   17) G. Taylor, Proc. R. Soc. London, Ser. A **280**, 383 (1964).

325   18) R. S. Legge and P. C. Lozano, J. Propul. Power **27**, 485 (2011).

326   19) D. Krejci, F. Mier-Hicks, R. Thomas, T. Haag, and P. Lozano, J. Spacecr. Rockets **54**, 447 (2017).

327   20) Y. Kitazawa, K. Ueno, and M. Watanabe, Chem. Rec. **18**, 391(2018).

328   21) F. Mier-Hicks and P. C. Lozano, J. Propul. Power **33**, 456 (2016).

329   22) D. G. Courtney, H. Shea, K. Dannenmayer, and A. Bulit, J. Spacecr. Rockets **55**, 54 (2018).

330   23) K. Nakagawa, T. Tsuchiya, and Y. Takao, Jpn. J. Appl. Phys. **56**, 06GN18 (2017).

331   24) F. Laermer and A. Schilp, U.S. Patent 5 501 893, Mar. 1996.

332   25) B. Wu, A. Kumar, and S. Pamarth, J. Appl. Phys. **108**, 051101-1 (2010).

333   26) C. Hedlund, H. O. Blom, and S. Berg, J. Vac. Sci. Technol. A **12**, 1962 (1994).

334   27) P. Lozano, M. Martínez-Sánchez, and J. M. Lopez-Urdiales, J. Colloid Interface Sci. **276**, 392 (2004).

335   28) N. Brikner and P. C. Lozano, Appl. Phys. Lett. **101**, 193504 (2012).

336   29) P. Lozano and M. Martínez-Sánchez, J. Colloid Interface Sci. **280**, 149 (2004).

337   30) F. A. Hill, E. V. Heubel, P. P. De Leon, and L. F. Velasquez-Garcia, J. Microelectromechanical Syst. **23**, 1237

338   (2014).

339   31) S. Dandavino, C. Ataman, C. N. Ryan, S. Chakraborty, D. Courtney, J. P. W. Stark, and H. Shea, J. Micromech.

340   Microeng. **24**, 075011 (2014).

341   32) D. G. Courtney and P. Lozano, Trans. Jpn. Soc. Aeronaut. Space Sci., Aerosp. Technol. Jpn. **8**, Pb_73 (2010).

342   33) D. M. Goebel and I. Katz, *Fundamentals of Electric Propulsion: Ion and Hall Thrusters* (Wiley, Hoboken,

343   NJ, 2008).

344   34) P. C. Lozano, J. Phys. D: Appl. Phys. **39**, 126 (2006).

345

346

347

List of Figure Captions

Fig. 1. Schematic of the externally-wetted electrospray thruster. High positive/negative voltage is applied to the emitter while the extractor electrode is grounded.

Fig. 2. Steps in the fabrication of small emitter arrays for preliminary tests. (a) First, a silicon wafer coated with a thick positive-type photoresist (PMER P-LA900PM) is exposed and developed. (b) Second, isotropic etching using SF_6 is performed to form the cone shape of the emitter tips. (c) After that, anisotropic etching using the Bosch process is performed to form the cylindrical shape of the emitters. (d) Finally, the photoresist is removed.

Fig. 3. (a) Image of a fabricated emitter chip with a single emitter. (b) SEM images of the single emitter and (c) the emitter tip. (d) Image of a fabricated emitter chip with nine emitters. (e) SEM images of the emitter at the center of the nine emitters and (f) the emitter on the far left of the nine emitters, 2.0 mm away from the center of the array.

Fig. 4. (a) Schematic of the experimental setup for the ion current measurements, where d and L are the gab distance between the emitter and extractor and the distance between the extractor and collector, respectively. Photographs of the extractor electrodes made of 0.08-mm thick stainless steel for (b) the single emitter and (c) the nine-emitter array.

368

369 **Fig. 5.** Collector current (solid circles), extractor current (open squares), and emitter current

370 (solid triangles) as functions of applied voltage for (a) the single emitter and (b) the nine-emitter

371 array.

372

373 **Fig. 6.** Modified process for fabricating the emitter arrays. (a) Silicon wafer coated with a thick

374 positive-type photoresist (AZ P4620) is exposed by a laser direct writing and (b) developed,

375 where the diameters of the resist for each emitter are adjusted according to the isotropic etching

376 rate. (c) Isotropic etching using SF_6 and (d) anisotropic etching using the Bosch process are

377 performed. (e) The resist is removed from the emitter tips. (f) Isotropic etching is repeated to

378 form sharp emitter tips.

379

380 **Fig. 7.** (a) Image of a fabricated chip with 361 emitters. (b) SEM images of the center of the

381 array, and of four emitters at distances of (c) 2.12 mm, (d) 3.54 mm, and (e) 5.66 mm from the

382 center of the array.

383

384 **Fig. 8.** (a) Image of a fabricated chip with 169 emitters. (b) SEM images of the emitter at the

385 center of the array, and emitters at distances of (c) 0.71 mm, (d) 2.12 mm, and (e) 3.54 mm

386 from the center of the array.

387

388   **Fig. 9.** (a) Image of a fabricated chip with 81-emitter chip. (b) SEM images of the emitter at the

389   center of the array, and emitters at distances of (c) 1.41 mm, (d) 2.82 mm, and (e) 5.66 mm

390   from the center of the array.

391

392   **Fig. 10.** (a) Image of an extractor electrode fabricated from a 525-µm thick silicon wafer. (b)

393   Schematic cross-sectional view of the extractor electrode taken along the white dotted line in

394   Fig. 10(a).

395

396   **Fig. 11**. Image of (a) the aligned emitter chip and the extractor electrode, and (b) the assembled

397   device.

398

399   **Fig. 12**. Collector current (solid circles), extractor current (open squares), and emitter current

400   (solid triangles) of (a) 81-emitter chip, (b) 169-emitter chip, and (c) 361-emitter chip as

401   functions of the applied voltage.

402

403   **Fig. 13**. Current per emitter for 361-(solid circles), 169-(solid triangles), and 81-(open squares)
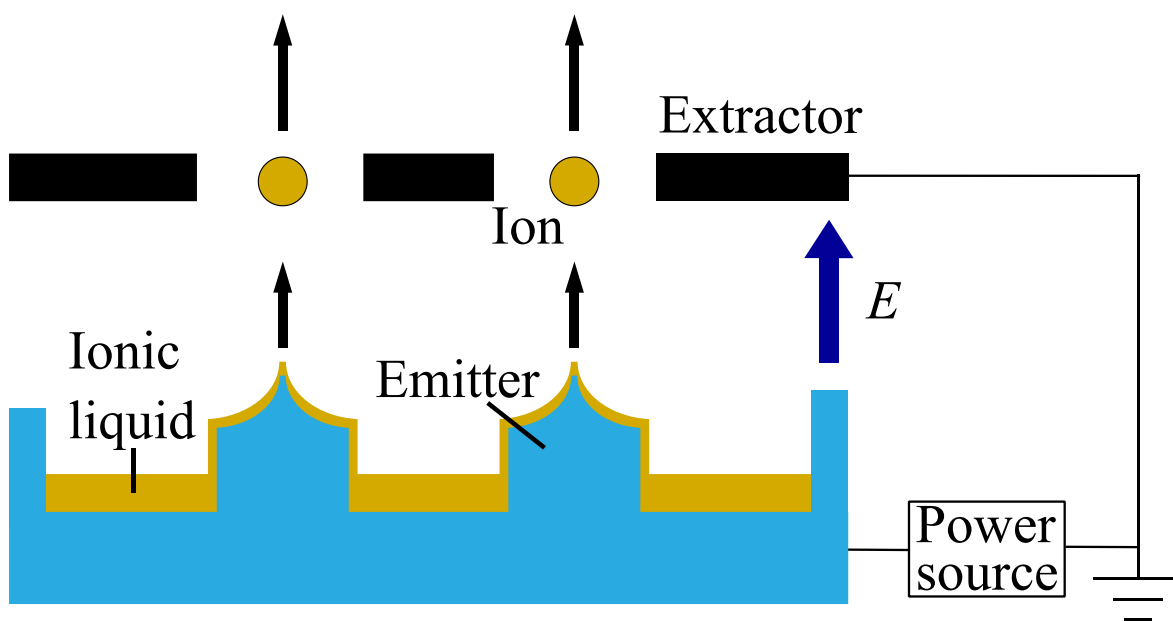
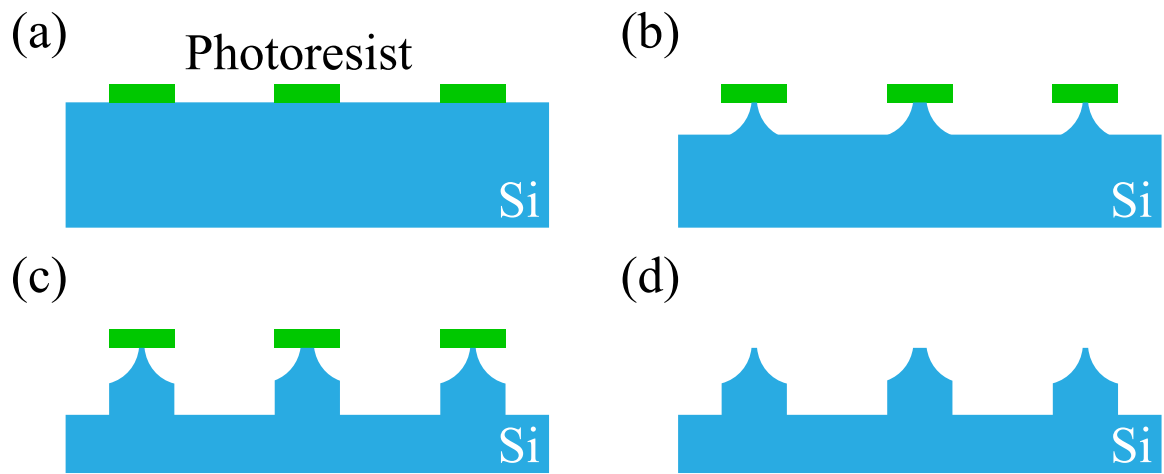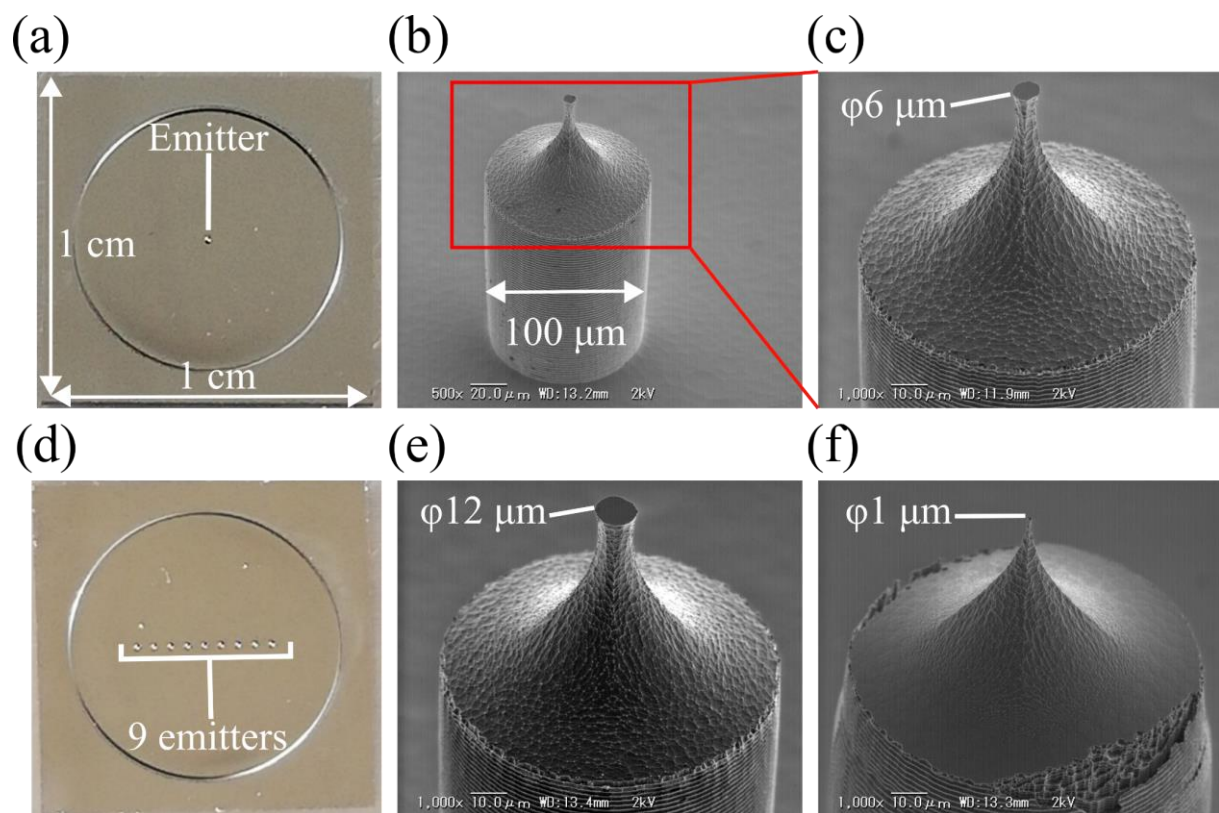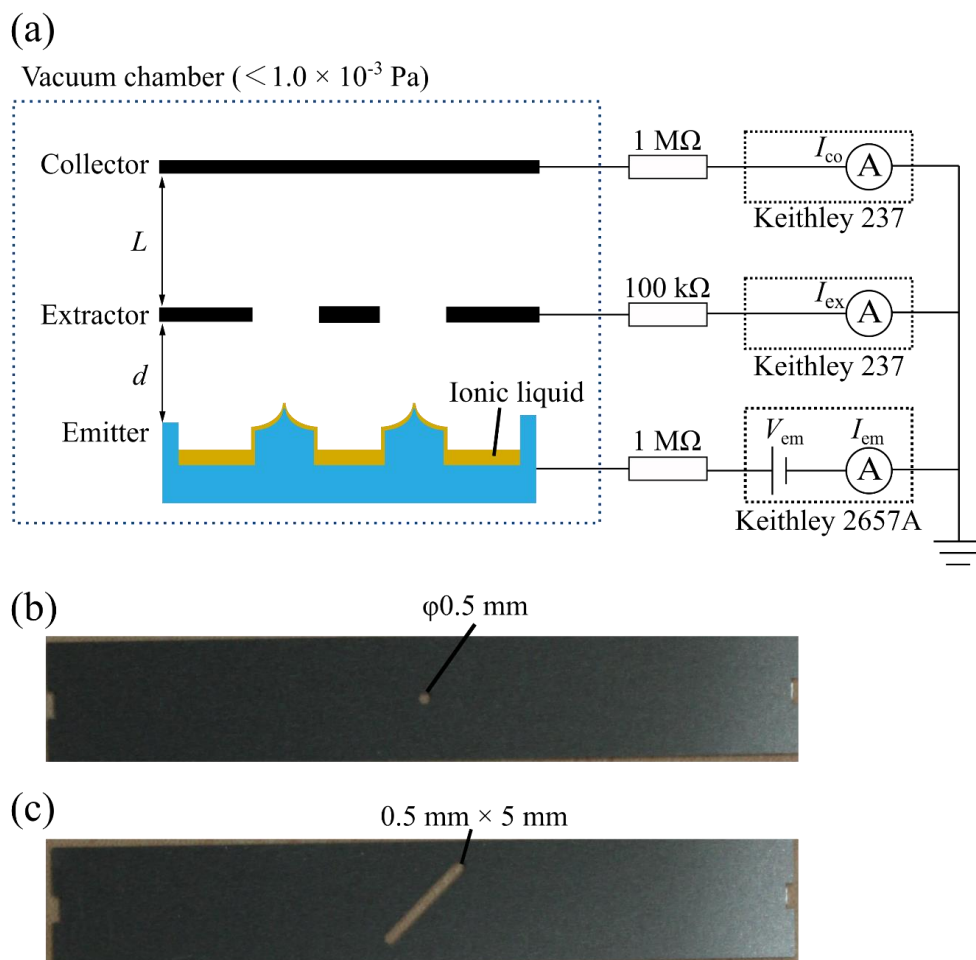404   emitter chips as functions of the applied voltage.

405

406

407

Extractor

Ion

Ionic
liquid

Emitter

E

Power
source

408

409

Fig. 1

22

(a) Photoresist

Si

(b)

Si

(c)

Si

(d)

Si

410

411

412

Fig. 2

23

(a) 1 cm / 1 cm / Emitter

(b) 100 μm / 500x 20.0 μm WD:13.2mm 2kV

(c) φ6 μm / 1,000x 10.0 μm WD:11.9mm 2kV

(d) 9 emitters

(e) φ12 μm / 1,000x 10.0 μm WD:13.4mm 2kV

(f) φ1 μm / 1,000x 10.0 μm WD:13.3mm 2kV

Fig. 3

(a)

Vacuum chamber ($<1.0 \times 10^{-3}$ Pa)

Collector

L

Extractor

d

Emitter

Ionic liquid

1 MΩ

I_{co} Ⓐ

Keithley 237

100 kΩ

I_{ex} Ⓐ

Keithley 237

1 MΩ

V_{em} I_{em} Ⓐ

Keithley 2657A

(b)

φ0.5 mm

(c)

0.5 mm × 5 mm

416

417   Fig. 4

418

25

(a)

(b)

419

420   Fig. 5

421

26

(a) Laser direct writing
Si

(b) Photoresist
Si

(c) Si

(d) Si

(e) Si

(f) Si

422

423

424

Fig. 6

(a) 13 mm / 13 mm / 18 mm / 18 mm

(b) Emitter tip / 95 μm / 310x 32.7μm WD:13.0mm 10kV

(c) 95 μm / 120x 83.3μm WD:14.6mm 10kV

(d) 95 μm / 120x 83.3μm WD:11.7mm 10kV

(e) 95 μm / 120x 83.3μm WD:16.5mm 10kV

425

426 Fig. 7

427

28

428

429   Fig. 8

430

431

432   Fig. 9

433

(a)

(b)

434

435   Fig. 10

436

(a) Emitter chip / Extractor electrode

(b) Electrical contact to emitter chip / Electrical contact to extractor electrode

437

438   Fig. 11

439

(a)

(b)

(c)

440

441   Fig. 12

442

33

443

444   Fig. 13